

Flight Dynamics & Control *Modal Analysis*

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Outline

- Similarity Transformations
- Eigenvalues & Eigenvectors
- Modal Coordinates
- Real & Complex Modes
- Lateral Dynamics: Dutch roll, roll and spiral modes
- Longitudinal Dynamics: Phugoid and Short-period modes



Similarity Transformations

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

$$x \in R^n, u \in R^m, y \in R^p$$

$$\dot{x} = Ax + bu$$

$$y = cx + du$$

Now consider the transformation to new states z , defined by

$$x = Tz \Leftrightarrow z = T^{-1}x$$

$$\begin{aligned} T\dot{z} &= ATz + Bu & \dot{z} &= T^{-1}ATz + T^{-1}Bu \\ y &= CTz + Du & y &= CTz + Du \end{aligned}$$

so that,

$$\begin{aligned} \dot{z} &= A^*z + B^*u, & A^* &= T^{-1}AT, B^* = T^{-1}B, C^* = CT, D^* = D \\ y &= C^*z + D^*u \end{aligned}$$

Eigenvalues & Eigenvectors

Consider the square $n \times n$ matrix A as a map from R^n to R^n , i.e.,

$$y = Ax$$

Does there exist a nontrivial input vector h , such that the output vector y , points in the same direction as h , i.e., $y = \lambda h$, λ where is some scalar?

$$Ah = \lambda h$$

Let's try to solve for h

$$[\lambda I - A]h = 0 \Rightarrow \text{A nontrivial solution exists iff } \det[\lambda I - A] = 0.$$

Characteristic polynomial $\phi(\lambda) \triangleq \det[\lambda I - A] = \lambda^n + \alpha_{n-1}\lambda^{n-1} + \cdots + \alpha_0$

$\phi(\lambda) = 0$ has n roots $\{\lambda_1, \dots, \lambda_n\}$, possibly complex, possibly repeated.

Choose one, say λ_i , $i \in \{\lambda_1, \dots, \lambda_n\}$ and find the corresponding h_i

$$[\lambda_i I - A]h_i = 0$$



Example 1 (distinct roots)

$$A = \begin{bmatrix} -5 & 3 \\ 3 & -5 \end{bmatrix} \Rightarrow \det[\lambda I - A] = \det \begin{bmatrix} \lambda + 5 & -3 \\ -3 & \lambda + 5 \end{bmatrix} = (\lambda + 5)^2 - 9$$
$$\phi(\lambda) = \lambda^2 + 10\lambda + 16 = (\lambda + 2)(\lambda + 8) \Rightarrow \boxed{\lambda_1 = -2, \lambda_2 = -8}$$

$$[-2I - A]h_1 = 0 \Rightarrow \begin{bmatrix} -2 + 5 & -3 \\ -3 & -2 + 5 \end{bmatrix} \begin{bmatrix} \eta_1 \\ \eta_2 \end{bmatrix} = 0$$

$$\begin{aligned} 3\eta_1 - 3\eta_2 &= 0 \\ -3\eta_1 + 3\eta_2 &= 0 \end{aligned} \Rightarrow \eta_1 = \eta_2 \Rightarrow \boxed{h_1 = \begin{bmatrix} 1 \\ 1 \end{bmatrix} \eta_2}$$

$$[-8I - A]h_1 \Rightarrow \begin{aligned} -3\eta_1 - 3\eta_2 &= 0 \\ -3\eta_1 - 3\eta_2 &= 0 \end{aligned} \Rightarrow \eta_1 = -\eta_2 \Rightarrow \boxed{h_2 = \begin{bmatrix} -1 \\ 1 \end{bmatrix} \eta_2}$$

Example 2 (repeated roots)

$$A = \begin{bmatrix} -5 & 0 \\ 0 & -5 \end{bmatrix} \Rightarrow \det[\lambda I - A] = \det \begin{bmatrix} \lambda + 5 & 0 \\ 0 & \lambda + 5 \end{bmatrix} = (\lambda + 5)^2$$

$$\phi(\lambda) = (\lambda + 5)(\lambda + 5) \Rightarrow \boxed{\lambda_1 = -5, \lambda_2 = -5}$$

$$[-5I - A]h_1 = 0 \Rightarrow \begin{bmatrix} -5+5 & 0 \\ 0 & -5+5 \end{bmatrix} \begin{bmatrix} \eta_1 \\ \eta_2 \end{bmatrix} = 0 \text{ satisfied for any } \eta_1, \eta_2$$

$$h_{1,2} = \begin{bmatrix} \eta_1 \\ \eta_2 \end{bmatrix} \quad \text{Choose} \quad \boxed{h_1 = \begin{bmatrix} 1 \\ 0 \end{bmatrix}, h_2 = \begin{bmatrix} 0 \\ 1 \end{bmatrix}}$$

Diagonal Form

eigen-system of A : $\begin{matrix} \lambda_1 & \lambda_2 & \cdots & \lambda_n \\ h_1 & h_2 & \cdots & h_n \end{matrix} \leftarrow \begin{matrix} \text{eigenvalues} \\ \text{independent eigenvectors} \end{matrix}$

$$T \triangleq [h_1 \ h_2 \ \cdots \ h_n]$$

$$\begin{aligned} \Rightarrow A^* &= [h_1 \ h_2 \ \cdots \ h_n]^{-1} A [h_1 \ h_2 \ \cdots \ h_n] \\ &= [h_1 \ h_2 \ \cdots \ h_n]^{-1} [Ah_1 \ Ah_2 \ \cdots \ Ah_n] \\ &= [h_1 \ h_2 \ \cdots \ h_n]^{-1} [\lambda_1 h_1 \ \lambda_2 h_2 \ \cdots \ \lambda_n h_n] \end{aligned}$$

$$= [h_1 \ h_2 \ \cdots \ h_n]^{-1} [h_1 \ h_2 \ \cdots \ h_n] \begin{bmatrix} \lambda_1 & 0 & \cdots & 0 \\ 0 & \lambda_2 & \ddots & \vdots \\ \vdots & \ddots & \ddots & 0 \\ 0 & \cdots & 0 & \lambda_n \end{bmatrix}$$

$$= \text{diag}(\lambda_1, \dots, \lambda_n)$$

$$\dot{z}_i = \lambda_i z_i + b_i^* u, i = 1, \dots, n$$

A decoupled system
of n 1st order ode's

Example

Define A

Compute eigensystem

Check similarity trans →

Use linear solve rather
than inv

```
>> A=[ 3 2 1;4 5 6;1 2 3];  
>> [V,D]=eig(A)  
V =  
-0.3482 -0.8581 0.4082  
-0.8704 0.1907 -0.8165  
-0.3482 0.4767 0.4082  
  
D =  
9.0000 0 0  
0 2.0000 0  
0 0 -0.0000
```

```
>> inv(V)*A*V  
ans =  
9.0000 -0.0000 -0.0000  
-0.0000 2.0000 -0.0000  
-0.0000 -0.0000 -0.0000
```

```
>> V\A*V  
ans =  
9.0000 -0.0000 0.0000  
-0.0000 2.0000 0  
-0.0000 -0.0000 0.0000
```



Modal Coordinates, 1

Consider the system in diagonal form. The z coordinates are referred to as 'modal coordinates'.

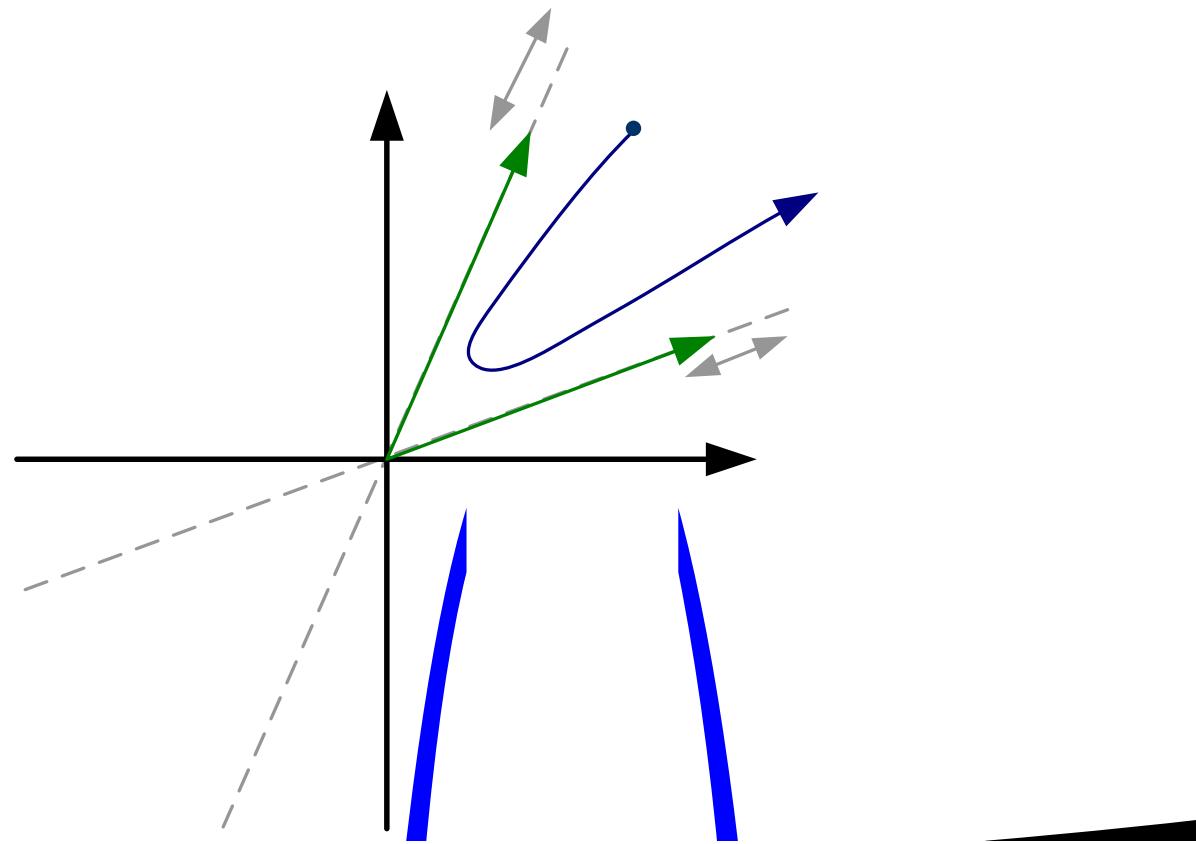
$$\dot{z}_i = \lambda_i z_i + \bar{b}_i u, i = 1, \dots, n$$

Suppose we solve these equations to obtain $z_i(t), i = 1, \dots, n$, then we can obtain the solution in the original coordinates via

$$x(t) = Tz(t) = [h_1 \quad h_2 \quad \dots \quad h_n] \begin{bmatrix} z_1(t) \\ z_2(t) \\ \vdots \\ z_n(t) \end{bmatrix} = h_1 z_1(t) + h_2 z_2(t) + \dots + h_n z_n(t)$$

Notice that each term $h_i z_i(t)$ describes a motion that takes place in the line defined by h_i .

Model Coordinates, 2



Modal Coordinates, 3

Suppose the system is unforced, $u(t) = 0$

The initial conditions for the modal coordinates are $z_0 = T^{-1}x_0$

The solution is

$$z_i(t) = e^{\lambda_i t} z_{0,i}, i = 1, \dots, n$$

$$x(t) = h_1 e^{\lambda_1 t} z_{0,1} + \dots + h_n e^{\lambda_n t} z_{0,n}$$

The modes are the vector time functions $h_i e^{\lambda_i t}$, h_i is the mode shape.

The solution is a linear combination of the modes.

If λ_i is real, so is h_i and the modal response is a simple exponential.

If there are n linearly independent eigenvectors, then the set of solutions

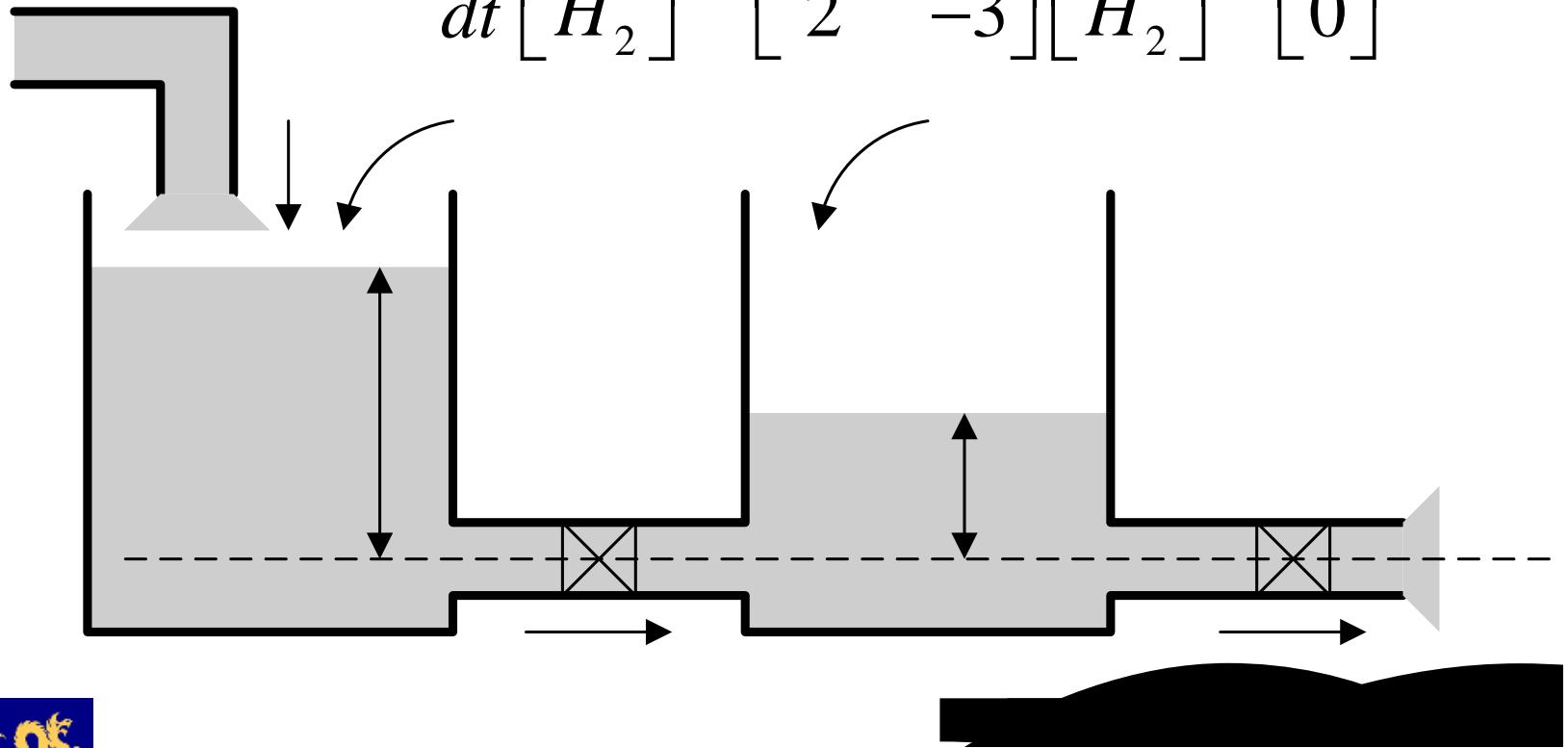
$$\{h_1 e^{\lambda_1 t}, \dots, h_n e^{\lambda_n t}\}$$

is called a fundamental set of solutions. Any solution to the homogeneous equation $\dot{x} = Ax$ is a linear combination of the fundamental solutions.

Example

$$A_1 = 1, A_2 = 1/2, R_1 = R_2 = 1$$

$$\frac{d}{dt} \begin{bmatrix} H_1 \\ H_2 \end{bmatrix} = \begin{bmatrix} -1 & 1 \\ 2 & -3 \end{bmatrix} \begin{bmatrix} H_1 \\ H_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 0 \end{bmatrix} Q_{in}$$



Example Cont'd

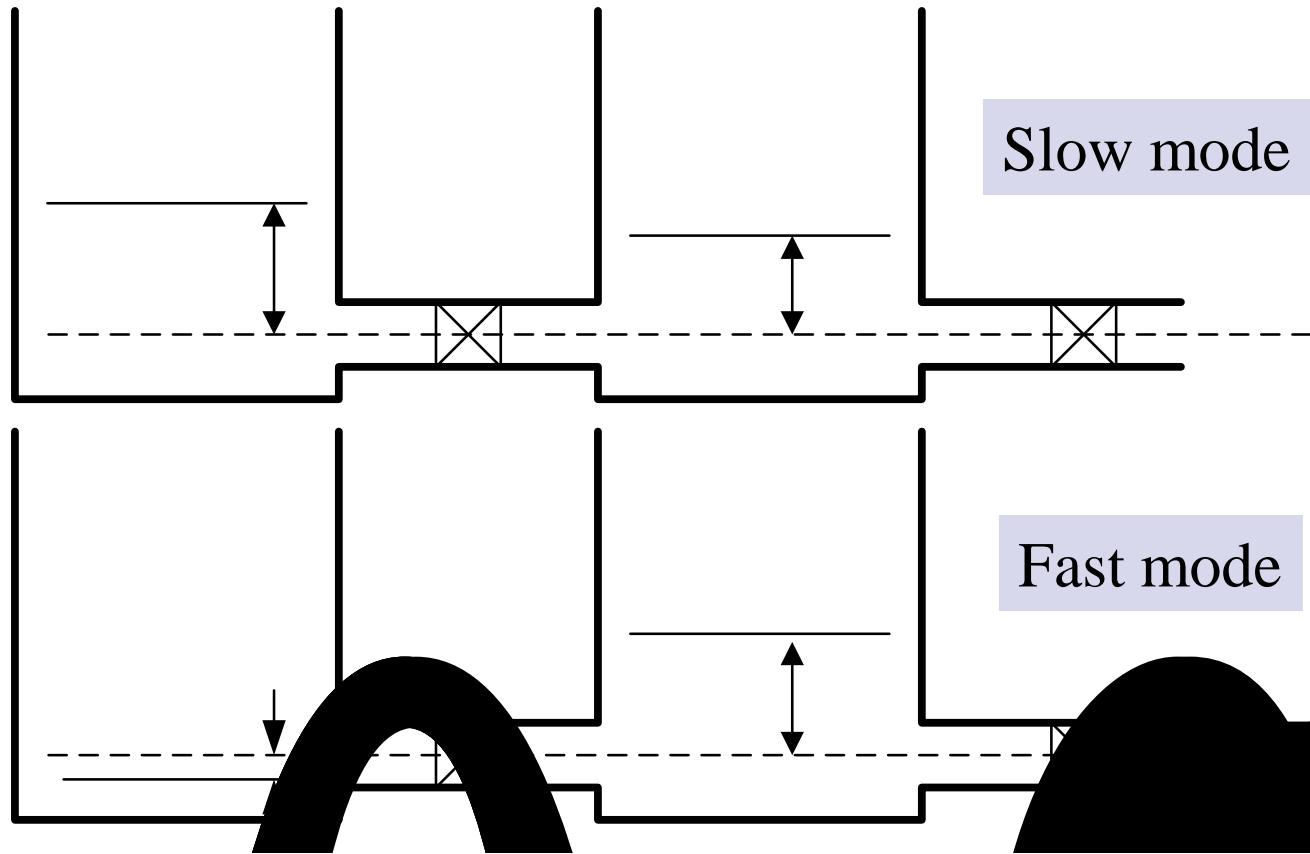
```
>> A=[ -1 1;2 -3];  
>> [E,V]=eig(A)  
E =  
    0.8069    -0.3437  
    0.5907     0.9391  
V =  
  
   -0.2679      0  
    0     -3.7321
```

slow mode: $\lambda_1 = -0.2679, h_1 = \begin{bmatrix} 0.8069 \\ 0.5907 \end{bmatrix}$

fast mode: $\lambda_2 = -3.7321, h_2 = \begin{bmatrix} -0.3437 \\ 0.9391 \end{bmatrix}$



Example, Cont'd



Complex Modes, 1

If the an eigenvalue λ_i is complex, the so is the eigenvector h_i .

To make things easier to interpret, we can construct real ones.

Suppose $\lambda_1 = \sigma + j\omega$ is a complex eigenvalue, and $\lambda_2 = \lambda_1^*$. Then

the corresponding eigenvectors are $h_1, h_2 = h_1^*$. Correspondly,

the two fundamental solutions are $h_1 e^{\lambda_1 t}, h_1^* e^{\lambda_1^* t}$. They are complex.

We will replace them with real ones.

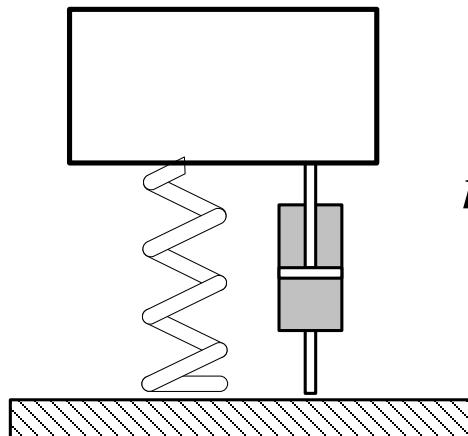
Complex Modes, 2

Define

$$\begin{aligned}x_1(t) &= \frac{1}{2} \left(h_1 e^{\lambda_1 t} + h_1^* e^{\lambda_1^* t} \right) \\&= \frac{1}{2} \left(\begin{aligned} &\left(h_{1,R} + j h_{1,I} \right) e^{\sigma t} (\cos \omega t + j \sin \omega t) \\&+ \left(h_{1,R} - j h_{1,I} \right) e^{\sigma t} (\cos \omega t - j \sin \omega t) \end{aligned} \right) \\&= h_{1,R} e^{\sigma t} \cos \omega t - h_{1,I} e^{\sigma t} \sin \omega t \\x_2(t) &= \frac{1}{2} \left(-h_1 e^{\lambda_1 t} + h_1^* e^{\lambda_1^* t} \right) = h_{1,R} e^{\sigma t} \sin \omega t + h_{1,I} e^{\sigma t} \cos \omega t\end{aligned}$$



Example



$$\dot{x} = v$$

$$m\dot{v} = -cv - kx$$

$$m = 1, k = 4, c = 1/2$$

$$\frac{d}{dt} \begin{bmatrix} x \\ v \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -4 & -0.5 \end{bmatrix} \begin{bmatrix} x \\ v \end{bmatrix}$$

$$h_{1,R} = \begin{bmatrix} 0.0559 \\ -0.8944 \end{bmatrix},$$

$$h_{1,I} = \begin{bmatrix} 0.4437 \\ 0 \end{bmatrix}$$

```
>> A=[ 0  1 ;-4 -0.5];
>> [E,V]=eig(A)
E =
    0.0559 + 0.4437i    0.0559 - 0.4437i
   -0.8944                  -0.8944
V =
   -0.2500 + 1.9843i      0
           0              -0.2500 - 1.9843i
```

Lateral Dynamics: Boeing 747 (cruise @ 40,000 ft)

```
>> A=[-0.0558 -0.997 0.0802 0.0415  
      0.598 -0.1150 -0.0318 0  
     -3.05 0.388 -0.465 0  
      0. 0.0805 1. 0];
```

Eigenvectors

$$\begin{bmatrix} \beta \\ r \\ p \\ \phi \end{bmatrix}$$

0.1994 - 0.1063i	0.1994 + 0.1063i	-0.0172	0.0067
-0.0780 - 0.1333i	-0.0780 + 0.1333i	-0.0118	0.0404
-0.0165 + 0.6668i	-0.0165 - 0.6668i	-0.4895	-0.0105
0.6930	0.6930	0.8717	0.9991

Eigenvalues

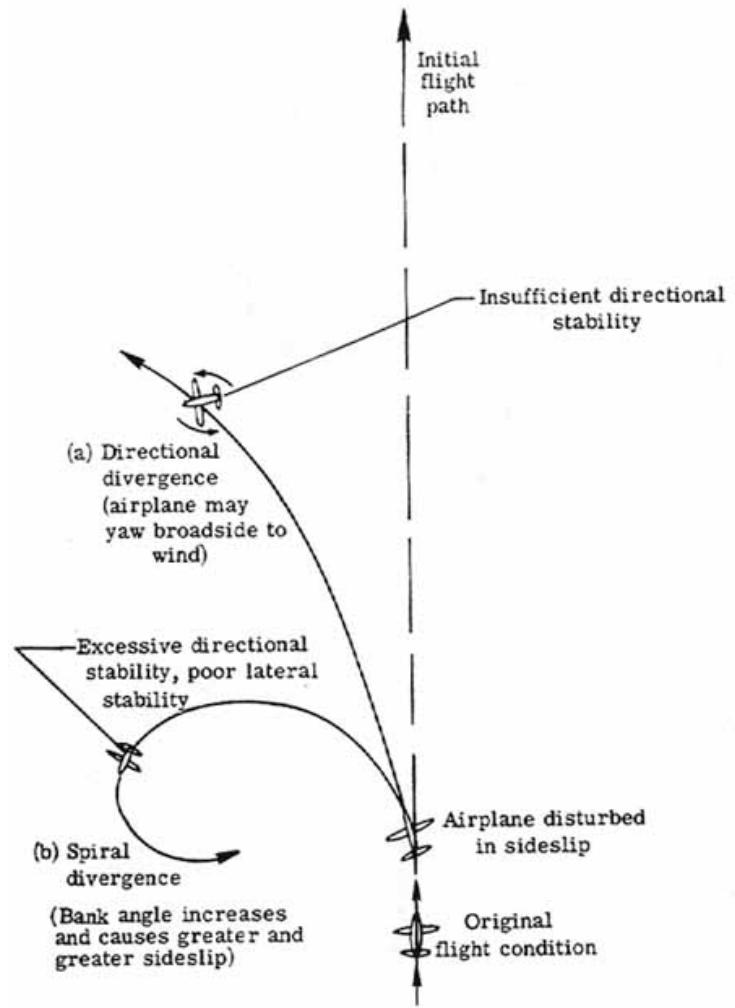
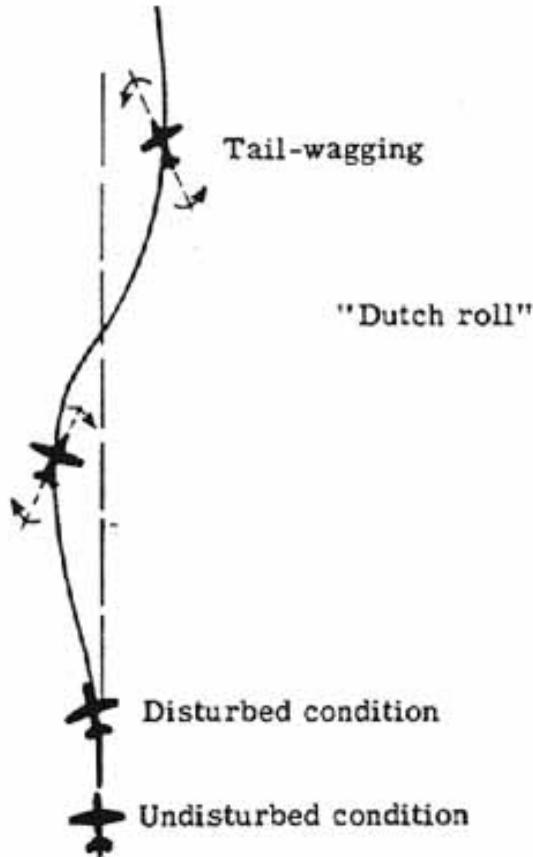
-0.0329 + 0.9467i -0.0329 - 0.9467i -0.5627 -0.0073

Dutch roll

Roll

Spiral

Dutch Roll & Spiral Modes



Longitudinal Dynamics: Boeing 747 (cruise@20,000ft)

```
>> A=[-0.00643 0.0263 0 -32.2  
      -0.0941 -0.624 820 0  
      -0.000222 -0.00153 -0.668 0  
      0 0 1 0];
```

$$\begin{bmatrix} u \\ v \\ q \\ \theta \end{bmatrix}$$

Eigenvectors

-0.0124 + 0.0040i	-0.0124 - 0.0040i	0.9894	0.9894
-0.9999	-0.9999	-0.1451 + 0.0005i	-0.1451 - 0.0005i
0.0000 - 0.0014i	0.0000 + 0.0014i	0.0000 - 0.0000i	0.0000 + 0.0000i
-0.0009 + 0.0005i	-0.0009 - 0.0005i	-0.0002 - 0.0003i	-0.0002 + 0.0003i

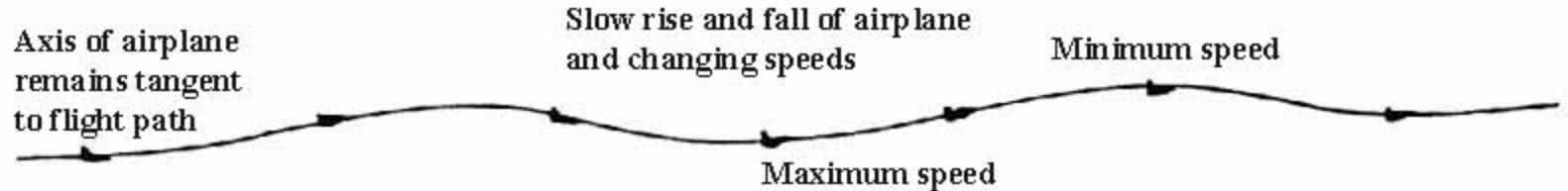
Eigenvalues

-0.6463 + 1.1211i	-0.6463 - 1.1211i	-0.0030 + 0.0098i	-0.0030 - 0.0098i
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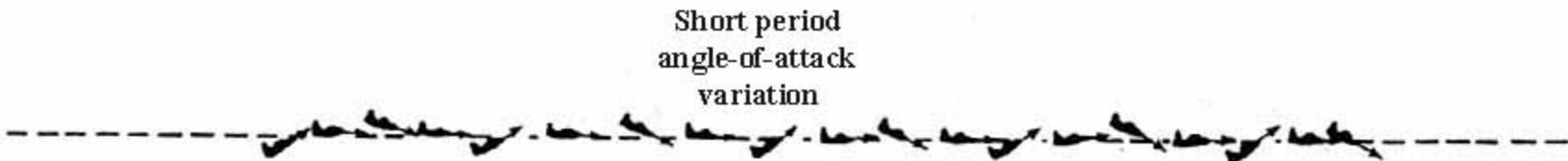
Short period

Phugoid

Phugoid and Short-period Modes



(a) Phugoid longitudinal oscillation.



(b) Short-period longitudinal oscillation.

Longitudinal Dynamics AFTI-16 (landing Configuration)

$$>> A = \begin{bmatrix} -0.0507 & -3.861 & 0 & -32.2 \\ -0.00117 & -0.5164 & 1 & 0 \\ -0.000129 & 1.4168 & -0.4932 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix}; \quad \begin{bmatrix} \Delta V \\ \Delta \alpha \\ q \\ \theta \end{bmatrix}$$

0.9943	0.9995	1.0000	1.0000
0.0633	-0.0142	0.0005 + 0.0003i	0.0005 - 0.0003i
-0.0740	-0.0165	0.0013 + 0.0002i	0.0013 - 0.0002i
0.0434	-0.0226	-0.0003 - 0.0064i	-0.0003 + 0.0064i
-1.7036	0.7310	-0.0438 + 0.2066i	-0.0438 - 0.2066i

Summary

- Similarity transformations
- Diagonalization using Eigenvectors
- Modes & Modal coordinates
- Interpreting behavior in terms of modes
- Converting complex modes into real ones
- Lateral – Dutch roll and spiral modes
- Longitudinal – phugoid and short period modes